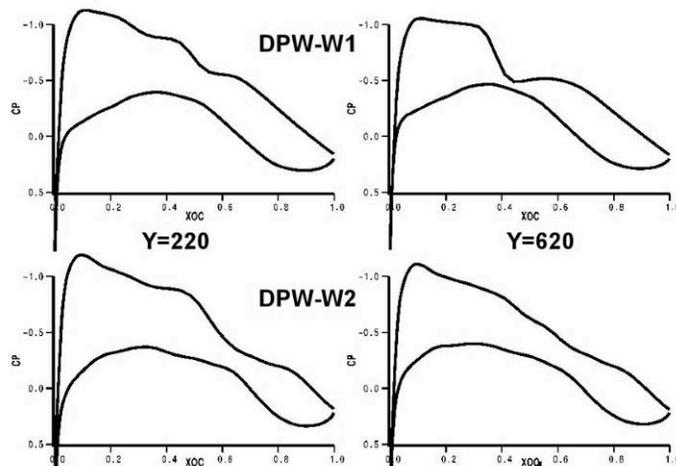


Case 2 DPW-W1/W2 Drag Prediction for the 3rd CFD Drag Prediction Workshop

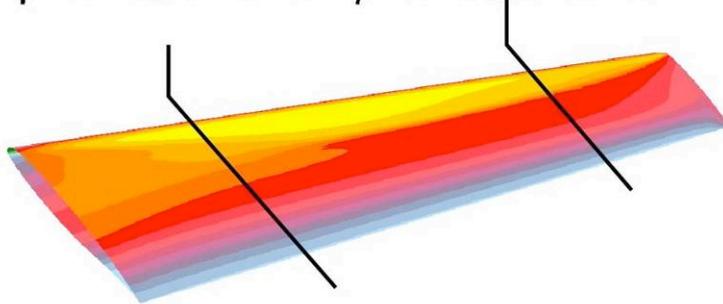
Mach=0.76

$\alpha=0.5$ deg



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Objective

Investigate the use of a “Production Navier-Stokes Analysis System” for CFD Drag Prediction

-Major interest is in the prediction of drag increments

-Use “standard” processes as much as possible

Acknowledgement

None of this work would have been possible without the considerable contributions of:

N. Jong Yu

Tsu-Yi Bernard Su

Tsong-Jhy Kao

Senthan Swaminathan

Moeljo Hong

Emanuel R Setiawan



CFL3D – Thin Layer Navier-Stokes Code

- Developed at NASA Langley (Jim Thomas, Kyle Anderson, Bob Biedron, Chris Rumsey, & ...)
- Finite volume
- Upwind biased and central difference
- Multigrid and mesh sequencing for acceleration
- Multiblock with 1-1 blocking, patched grid, and overlap-grid
- Numerous turbulence models
 - Spalart-Almaras SA Model
 - Menter's $k-\omega$ SST Model
- Time accurate with dual-time stepping
- Runs efficiently on parallel machines through MPI

Run with ICEM Generated Structured Grids



CFD++ – Unstructured Grid Navier-Stokes Code

- Developed by Metacomp Technologies
- Unified grid, unified physics and advanced numerical discretization and solution framework.
- Finite volume
- Upwind biased
- Multigrid for acceleration
- Arbitrary elements and has overset capabilities.
- Choice of turbulence models
 - Spalart-Almaras SA Model
 - $k-\varepsilon$ -Rt Model
- Time accurate with dual-time stepping
- Runs efficiently on parallel machines through MPI

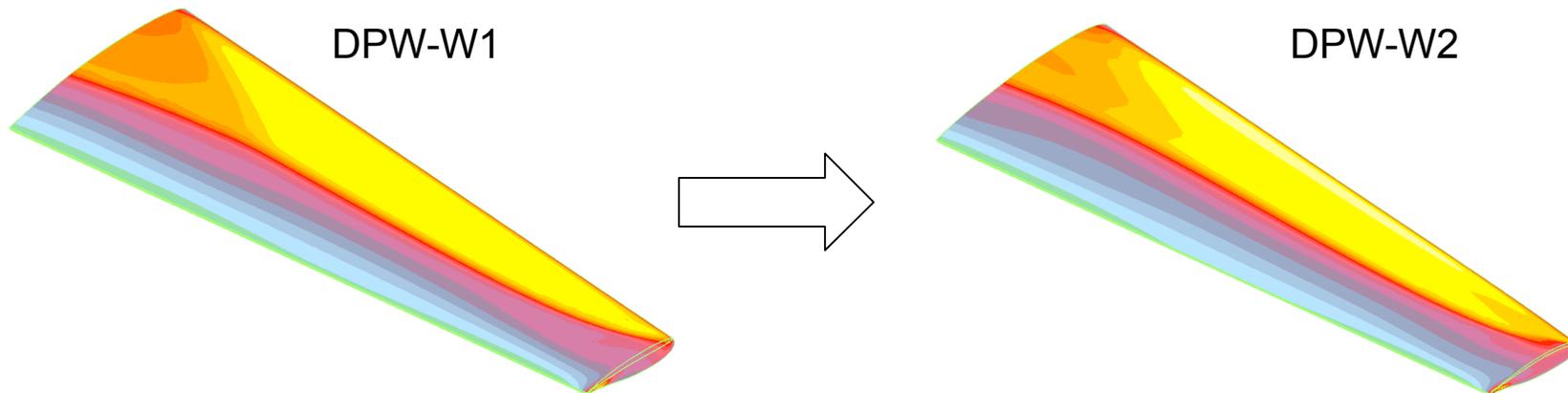
Limited runs with:

- ICEM Structured Grids
- Cessna Unstructured Grids

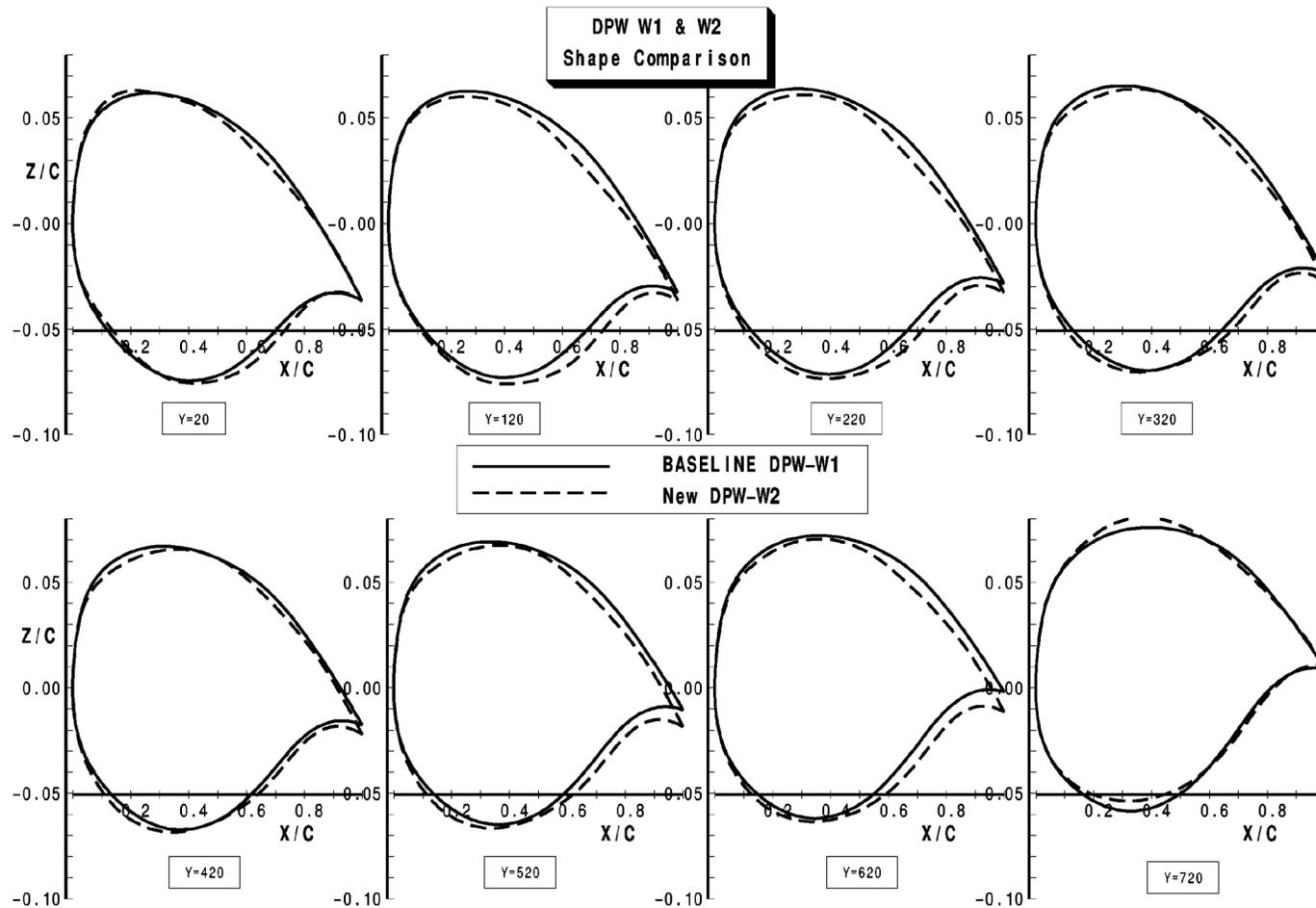
DPW-W2 Creation

Objective: Create a companion wing to DPW-W1 for drag increment prediction

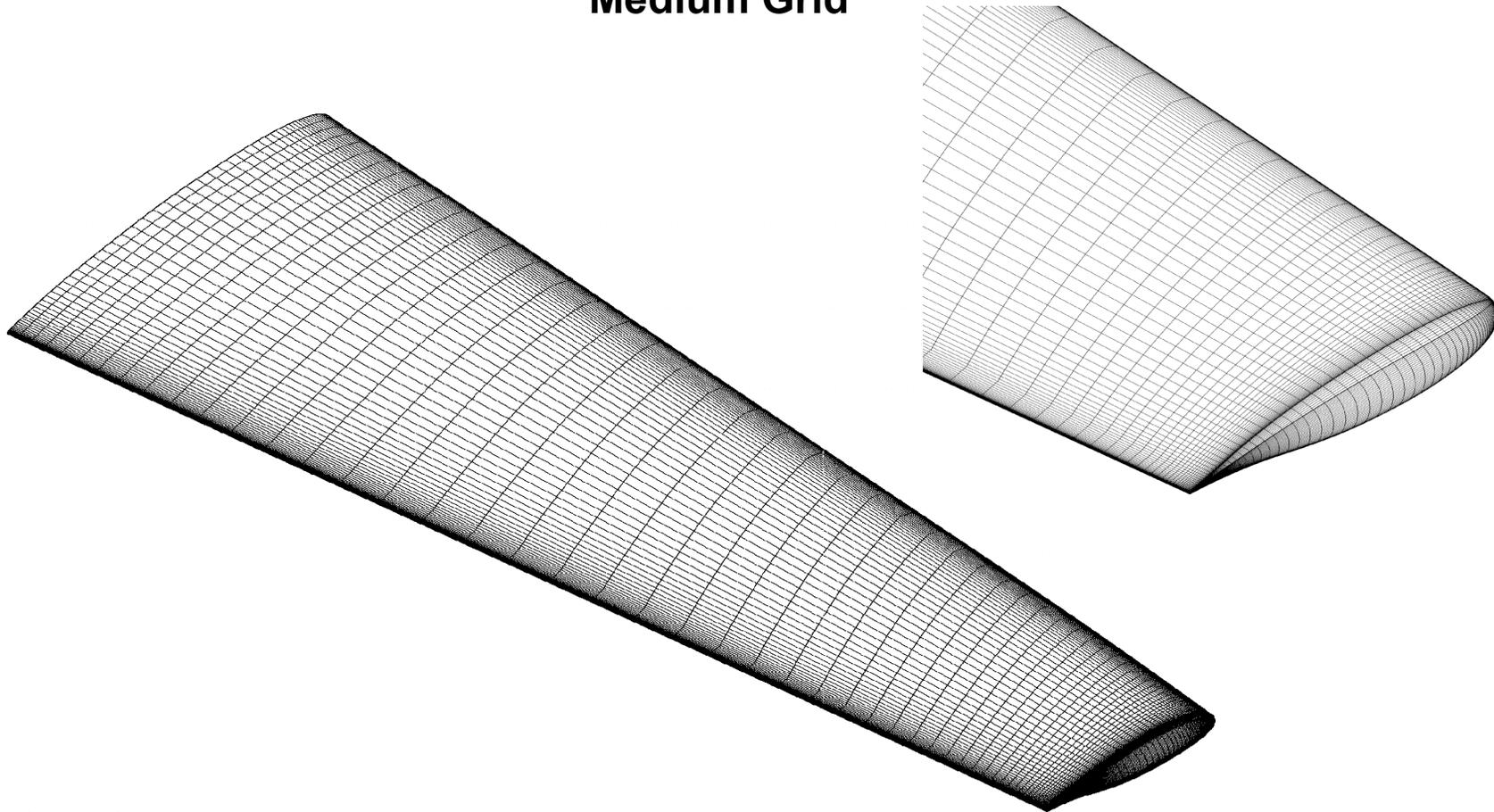
- Maintain the same planform and thickness
- Use optimization to change camber and twist
 - TRANAIR single-point optimization
 - Sequential Quadratic Programming
 - Linear Constraints
 - Nonlinear Objectives
- Minimize drag at a specified lift
- Variables: 5 camber variables + twist + shear @ 7 spanwise locations



DPW-W1/W2 Shape Comparisons



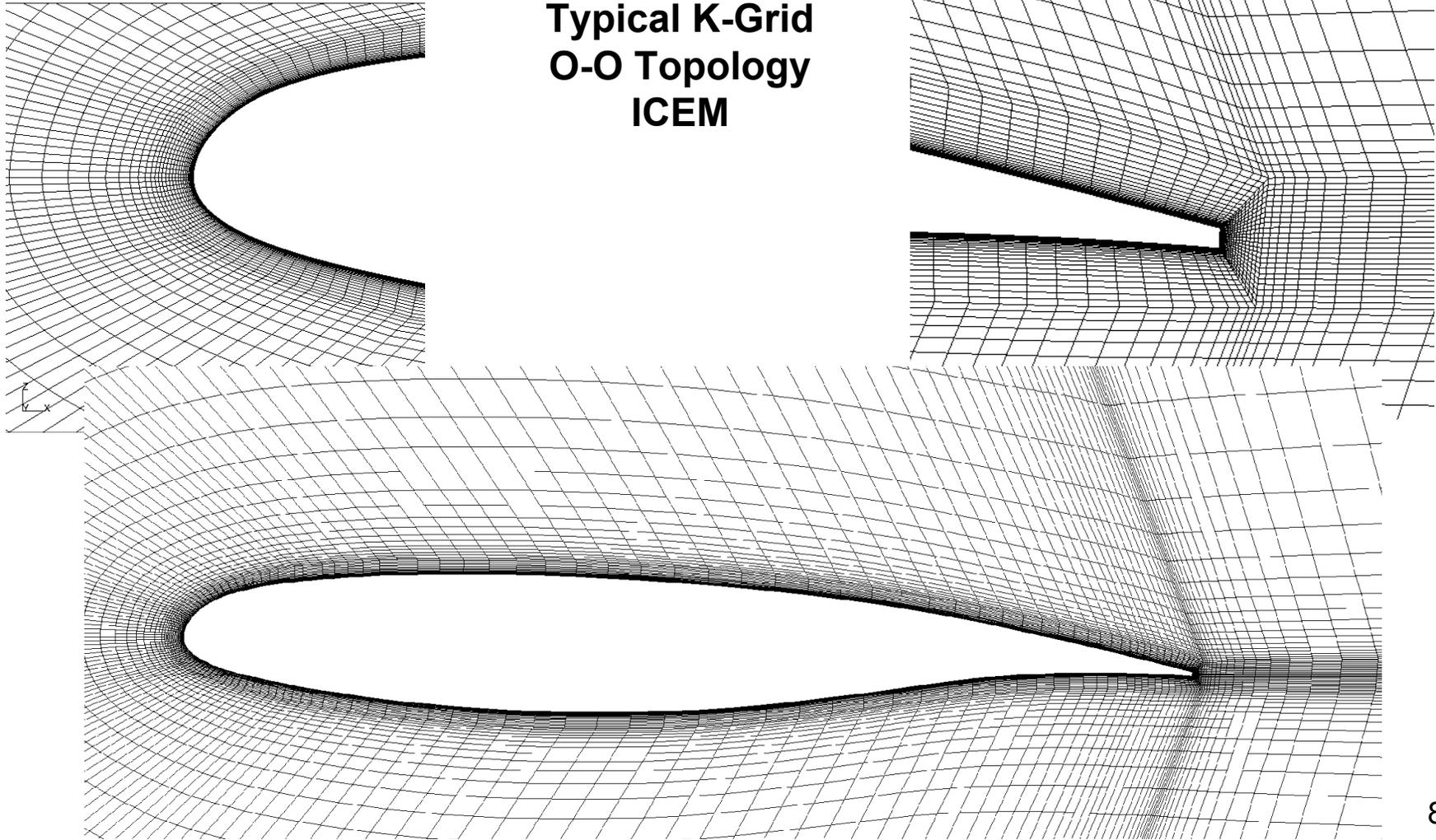
DPW-W1/W2 ICEM Structured Grids Medium Grid



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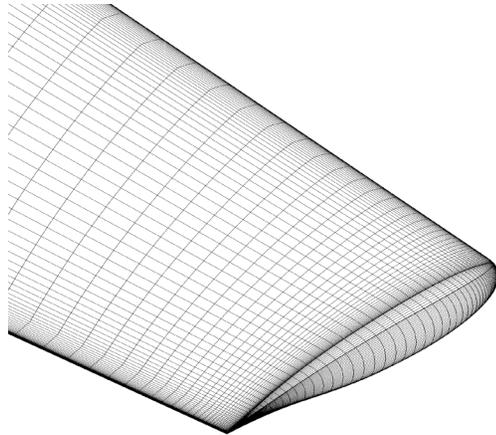
DPW-W1/W2
Typical K-Grid
O-O Topology
ICEM



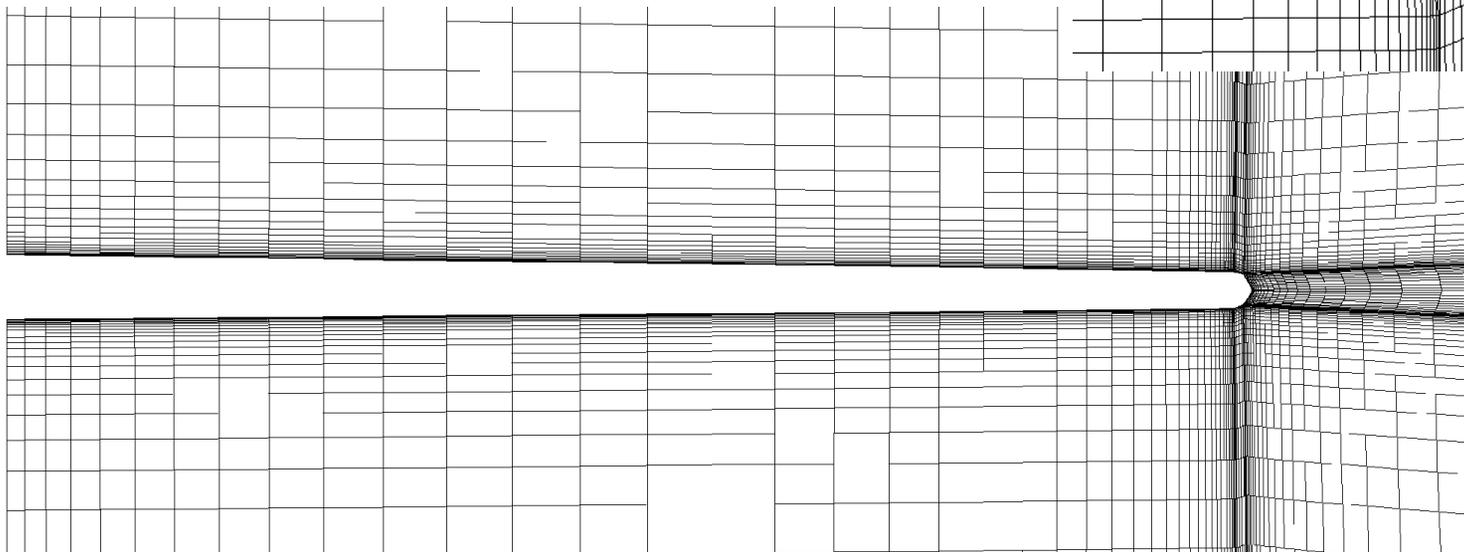
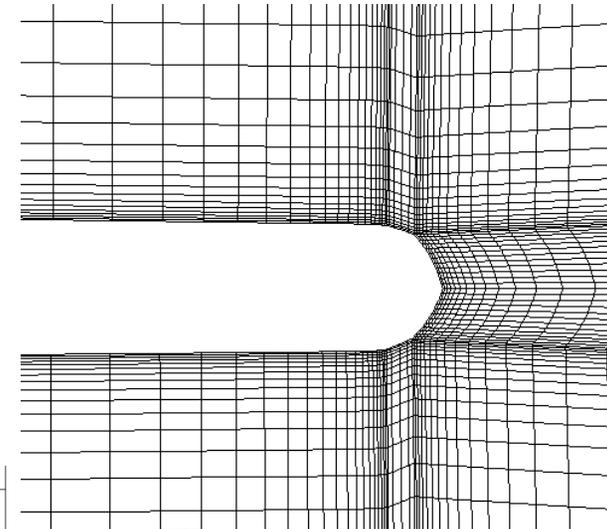


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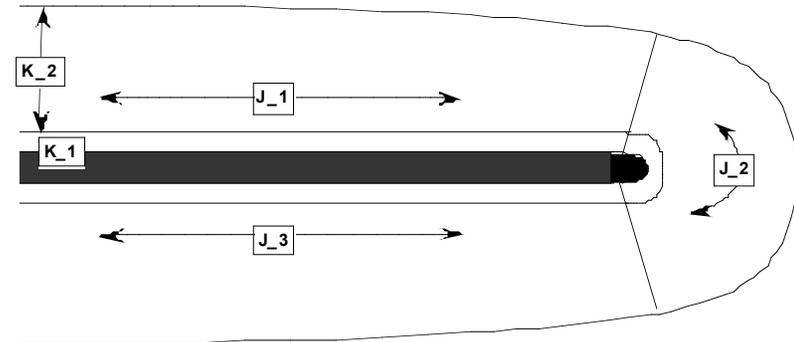
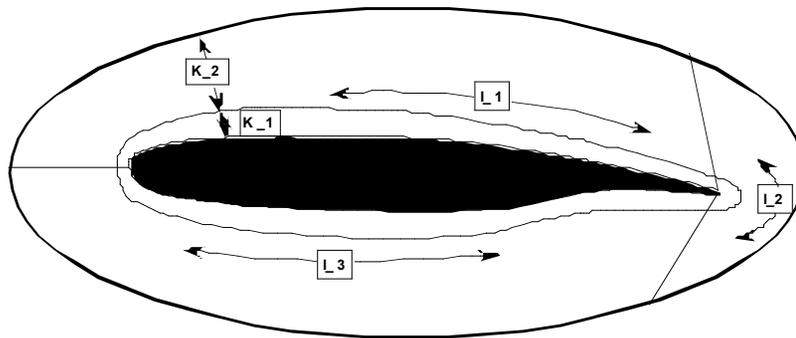


DPW-W1/W2
Typical I-Grid
O Topology
ICEM



Structured Multi-Block DPW-W1/W2 Grids

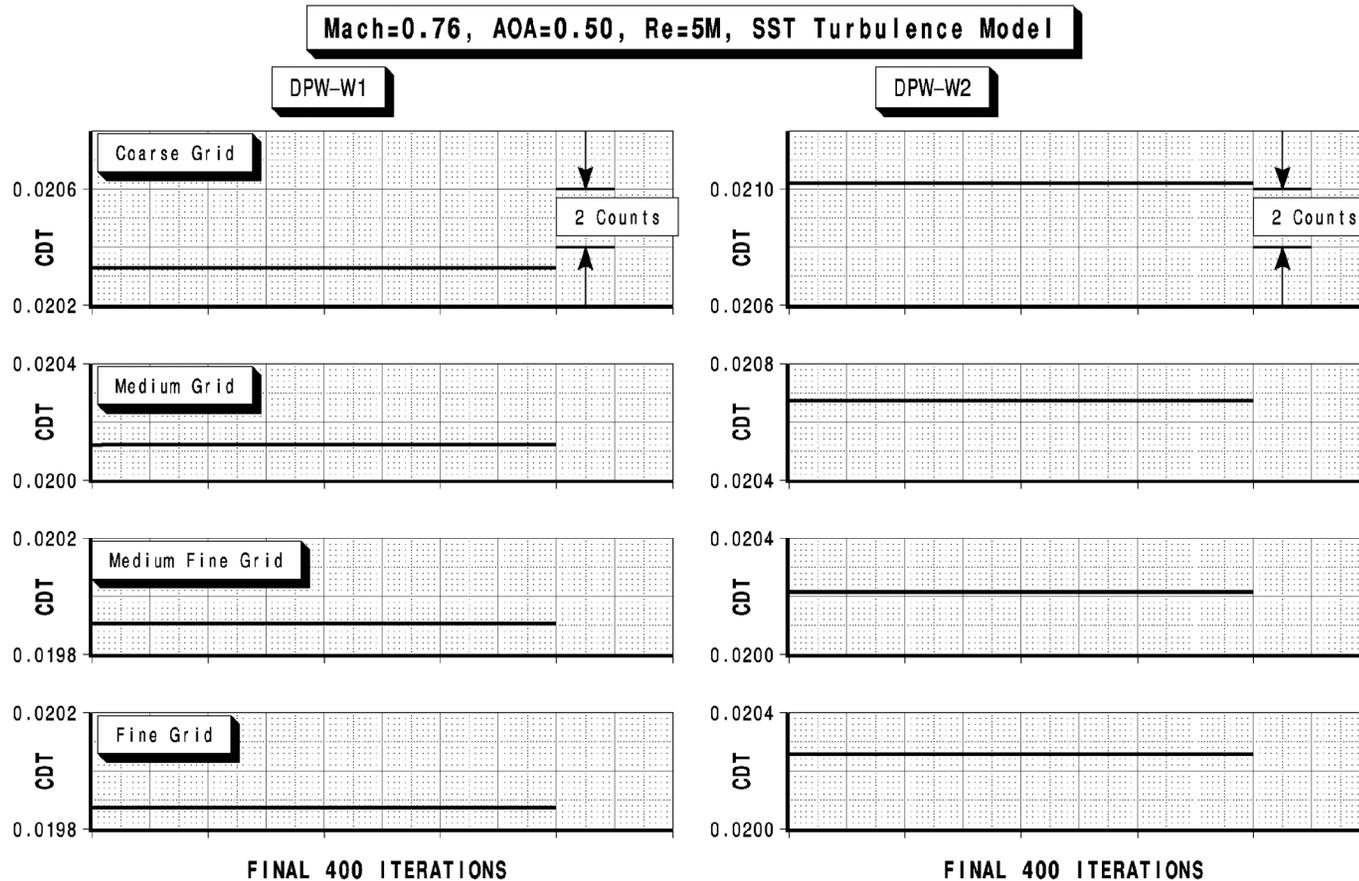
Constructed with ICEM



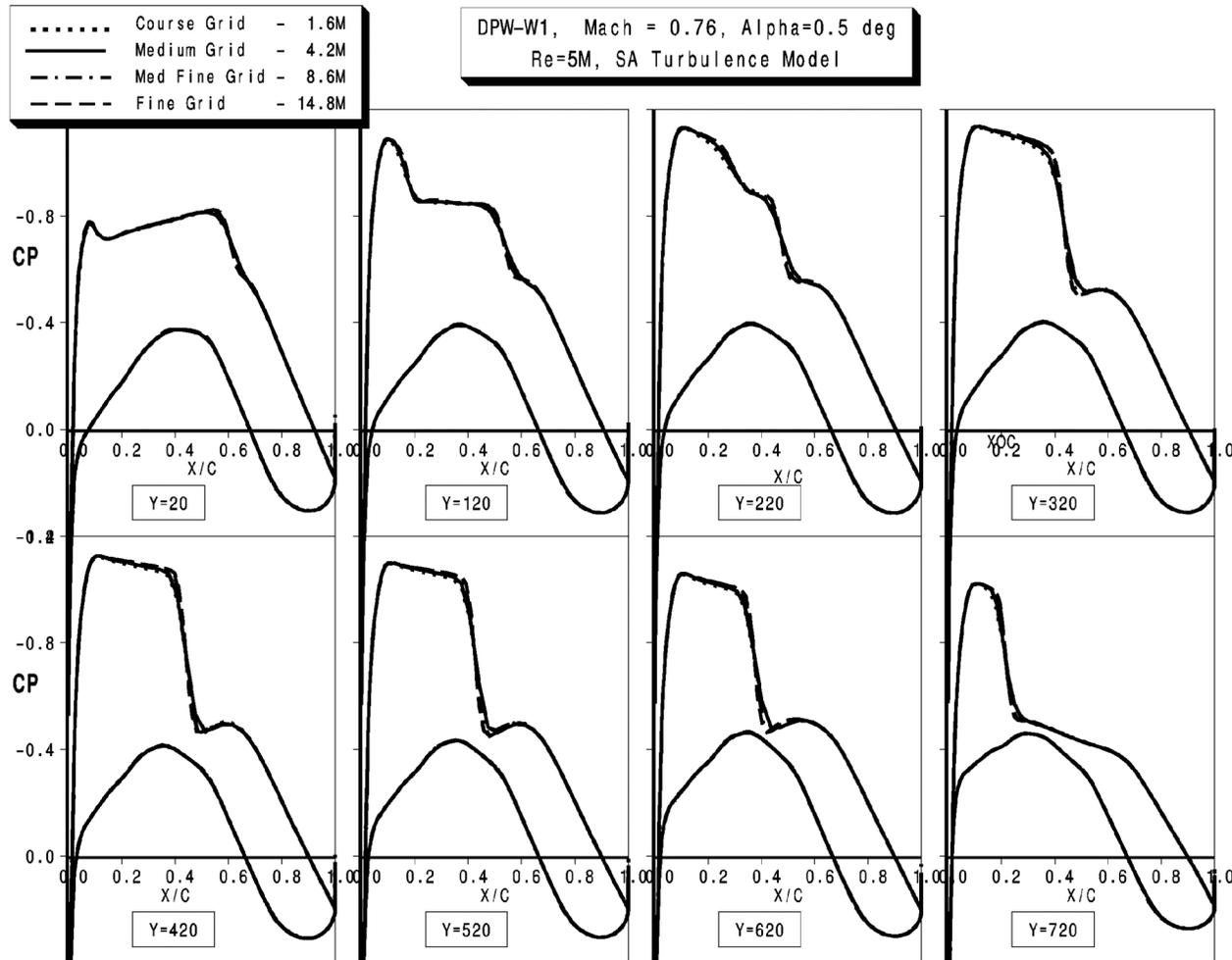
	I_1	I-2	I_3	J_1	J_2	J_3	K_1	K_2	Total Grid Size
Coarse	73	25	73	49	25	49	33	49	1.60E+06
Medium	81	33	81	73	33	73	49	73	4.20E+06
Medium Fine	121	49	121	73	49	73	65	97	8.60E+06
Fine	145	49	145	105	49	105	73	105	1.47E+07

Gridding Guidelines not met - Grids were not uniformly refined!

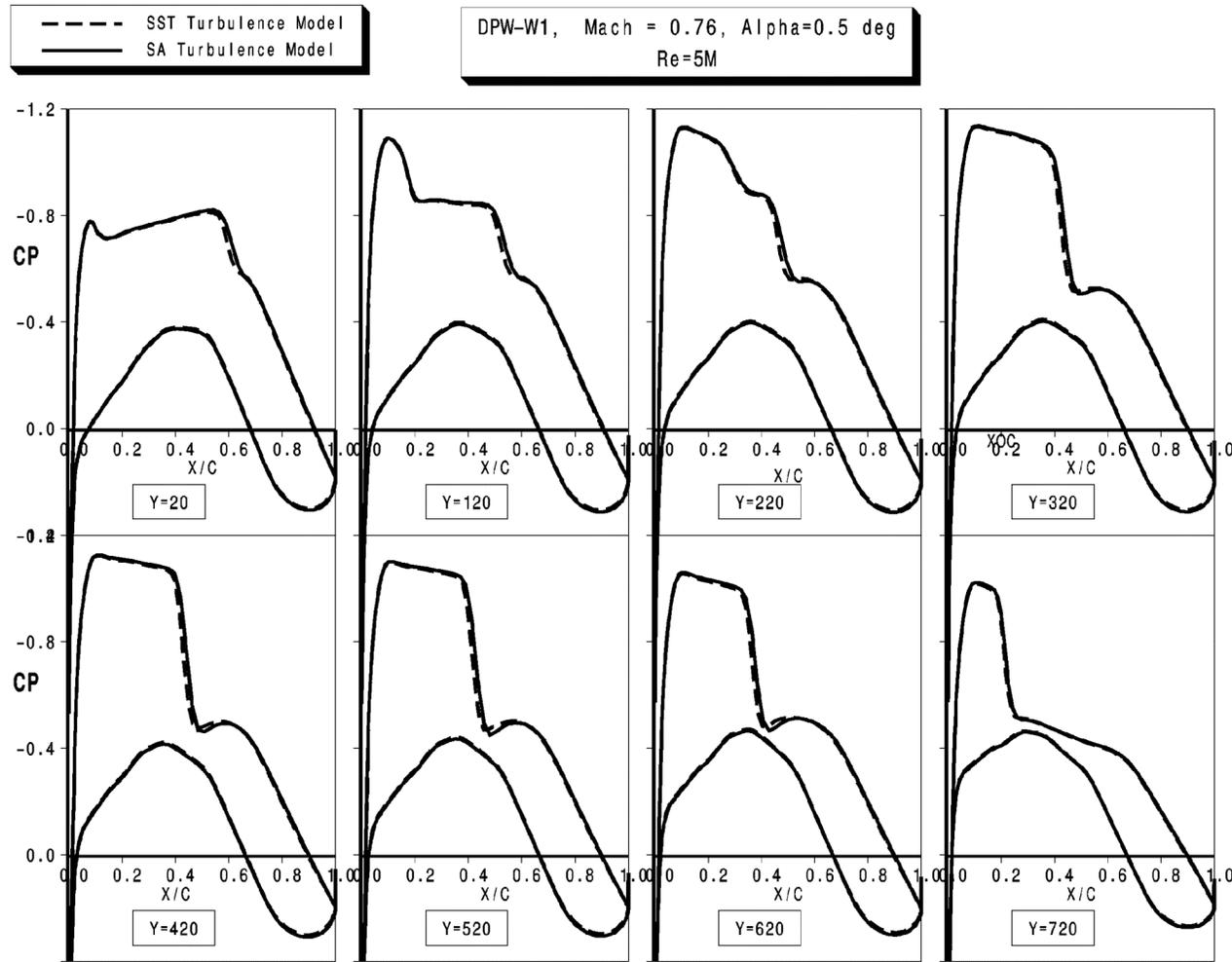
DPW-W1/W2 – Drag Convergence – CFL3D



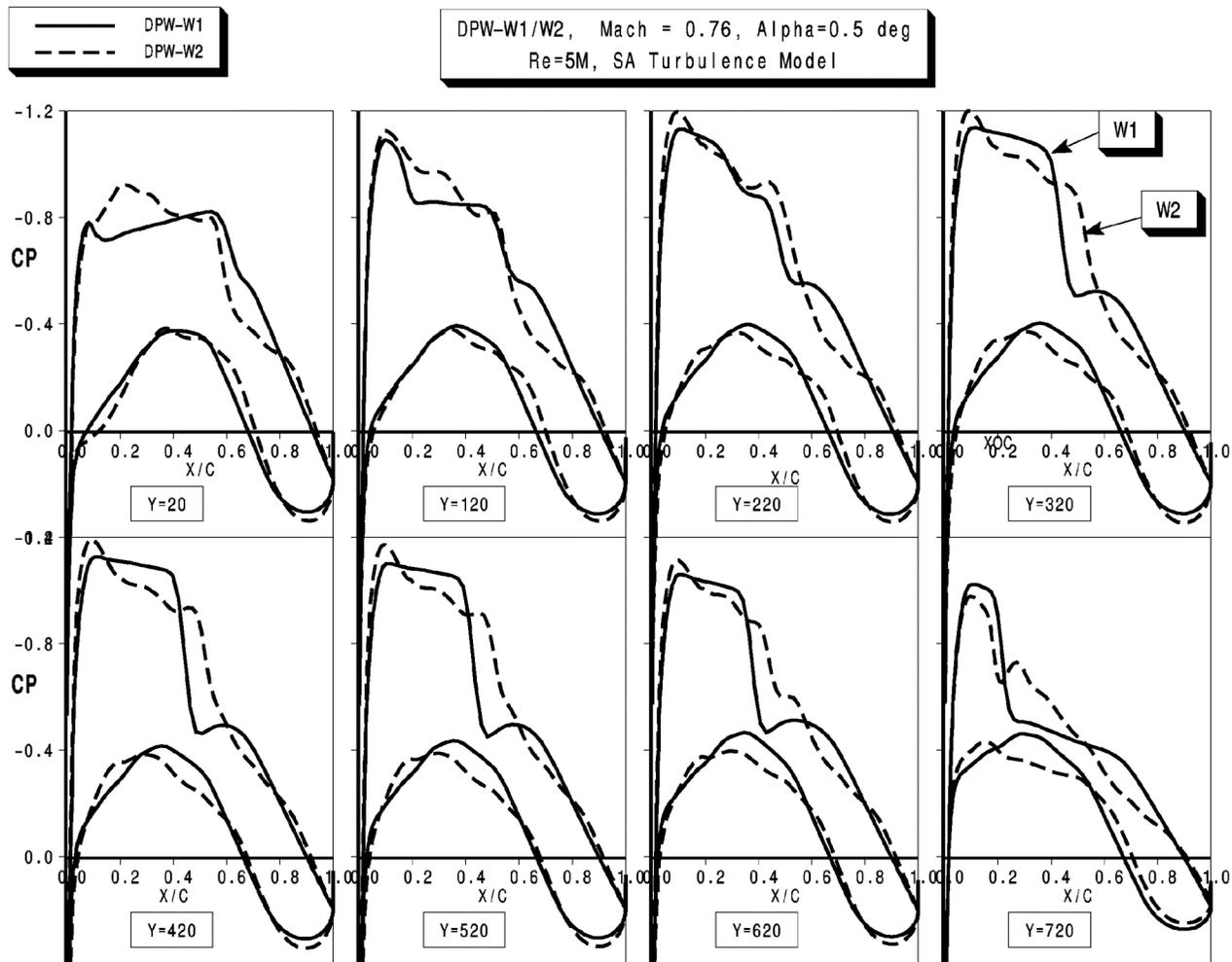
DPW-W1 – Wing Cp's – Grid Convergence – CFL3D



DPW-W1 – Wing Cp's – Turbulence Modeling Effects – CFL3D

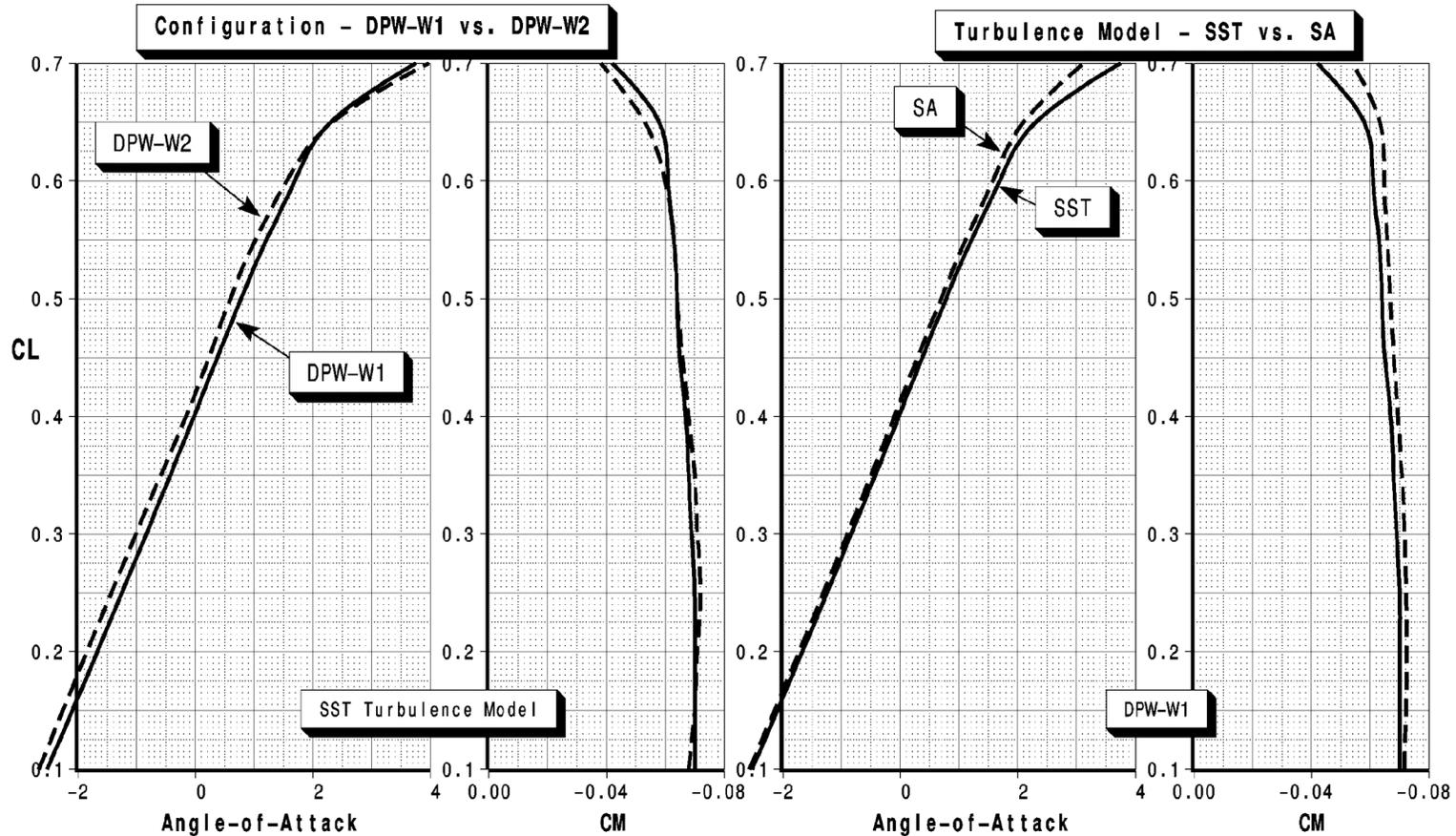


DPW-W1 vs W2 – Wing Cp's – CFL3D

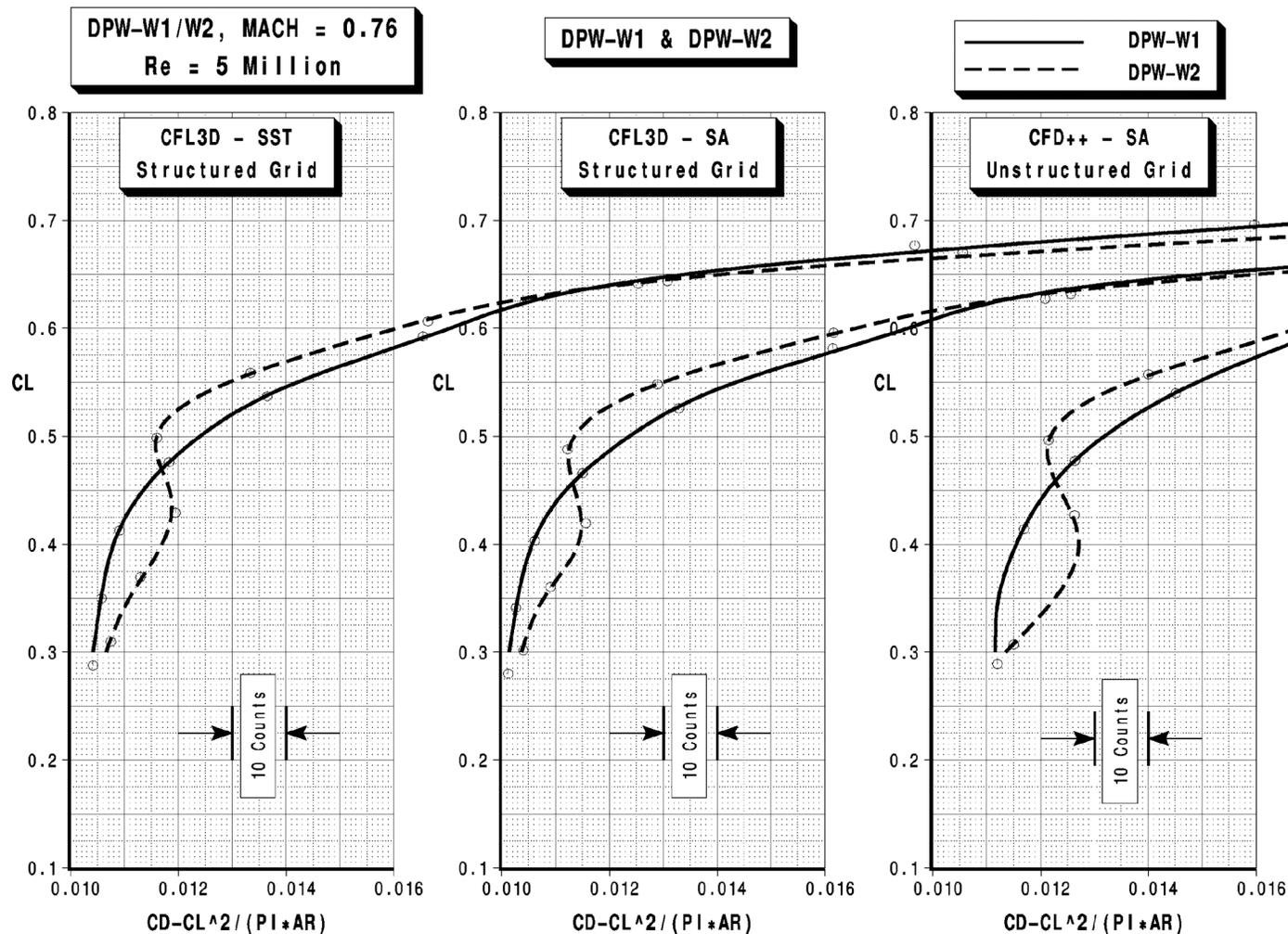


DPW-W1 / W2 – Lift and Pitching Moment – CFL3D

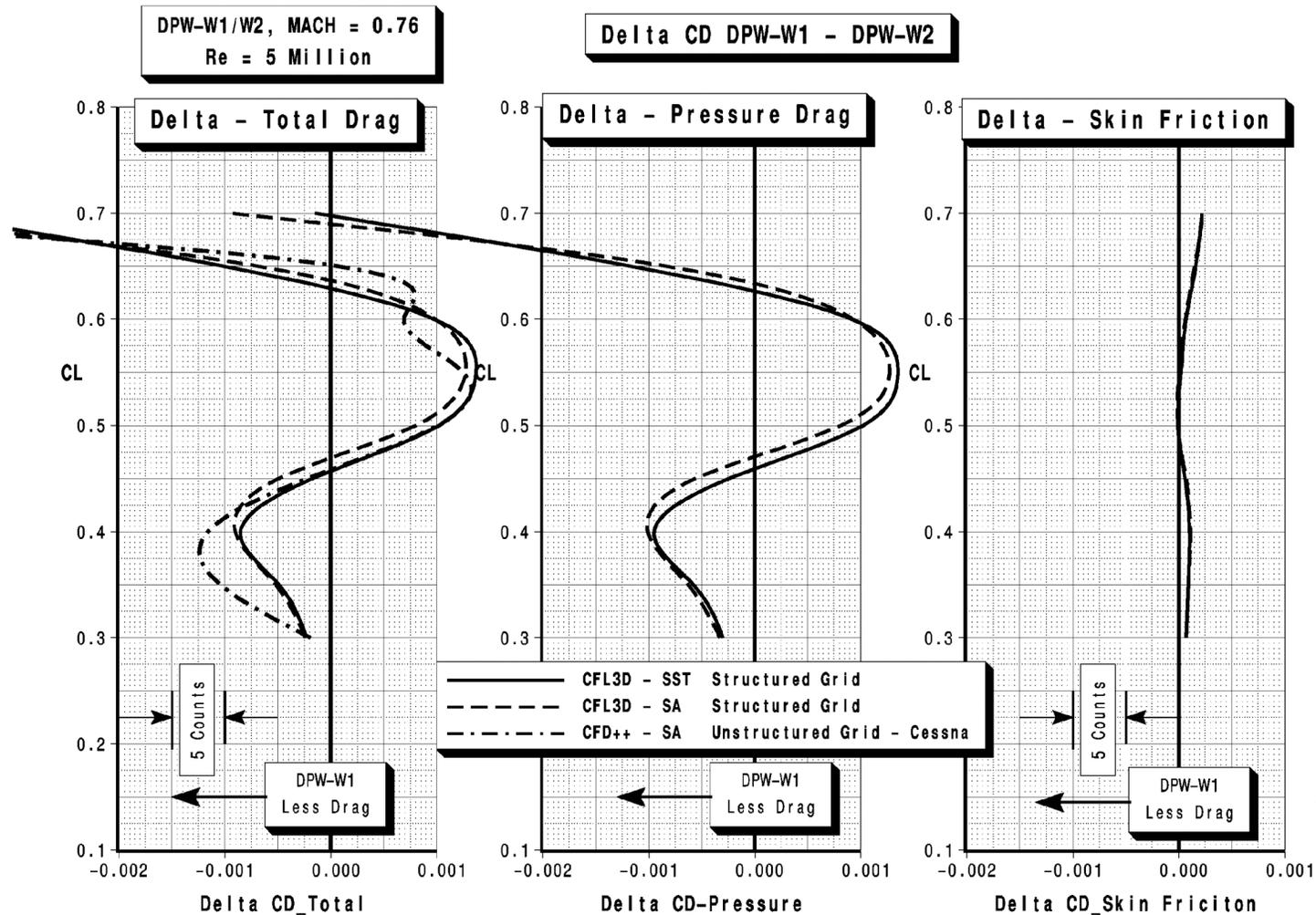
DPW-W1/W2, MACH = 0.76
Re = 5 Million



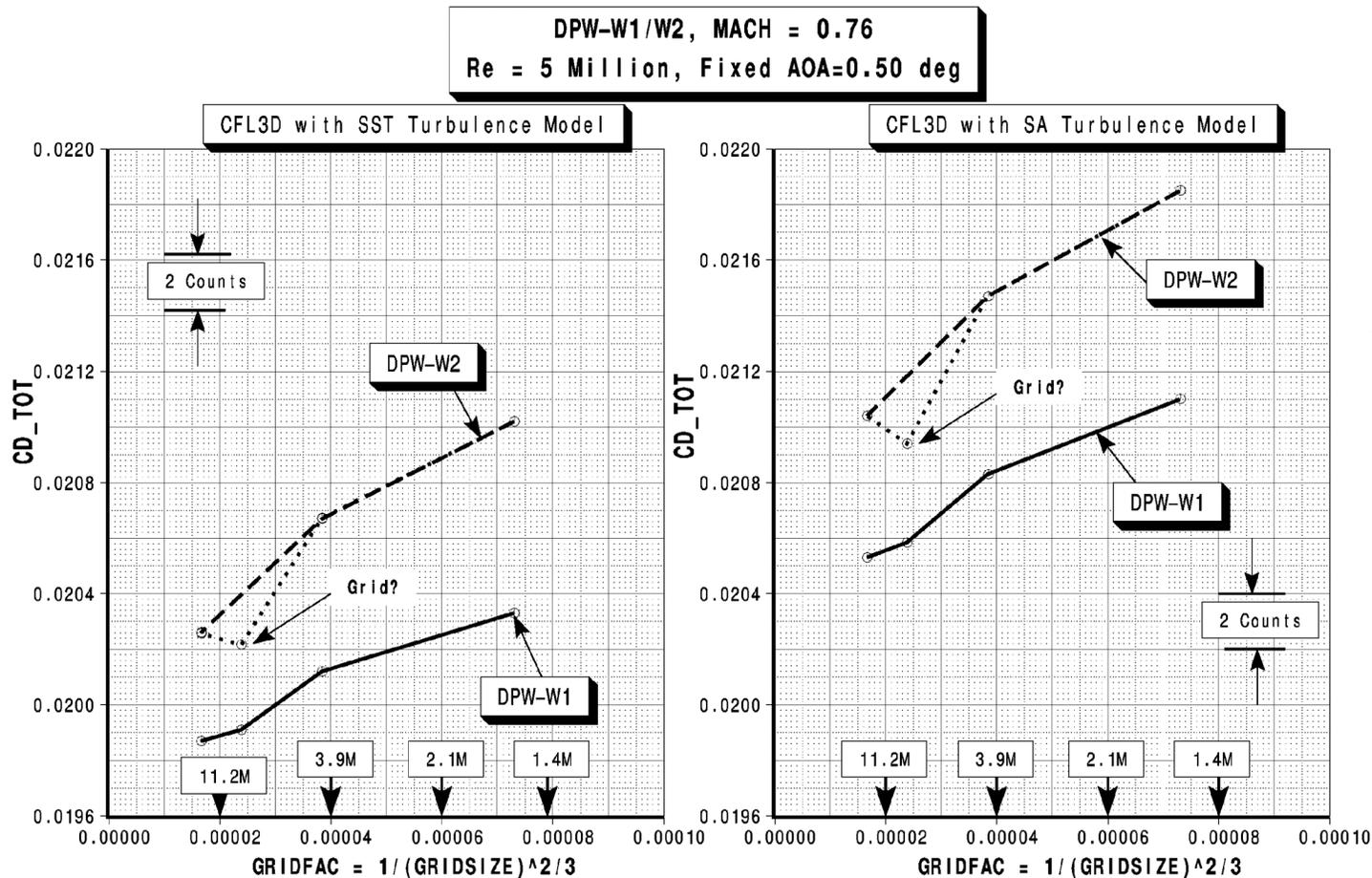
DPW-W1 / W2 – Polar Shape - Code/Turbulence Modeling



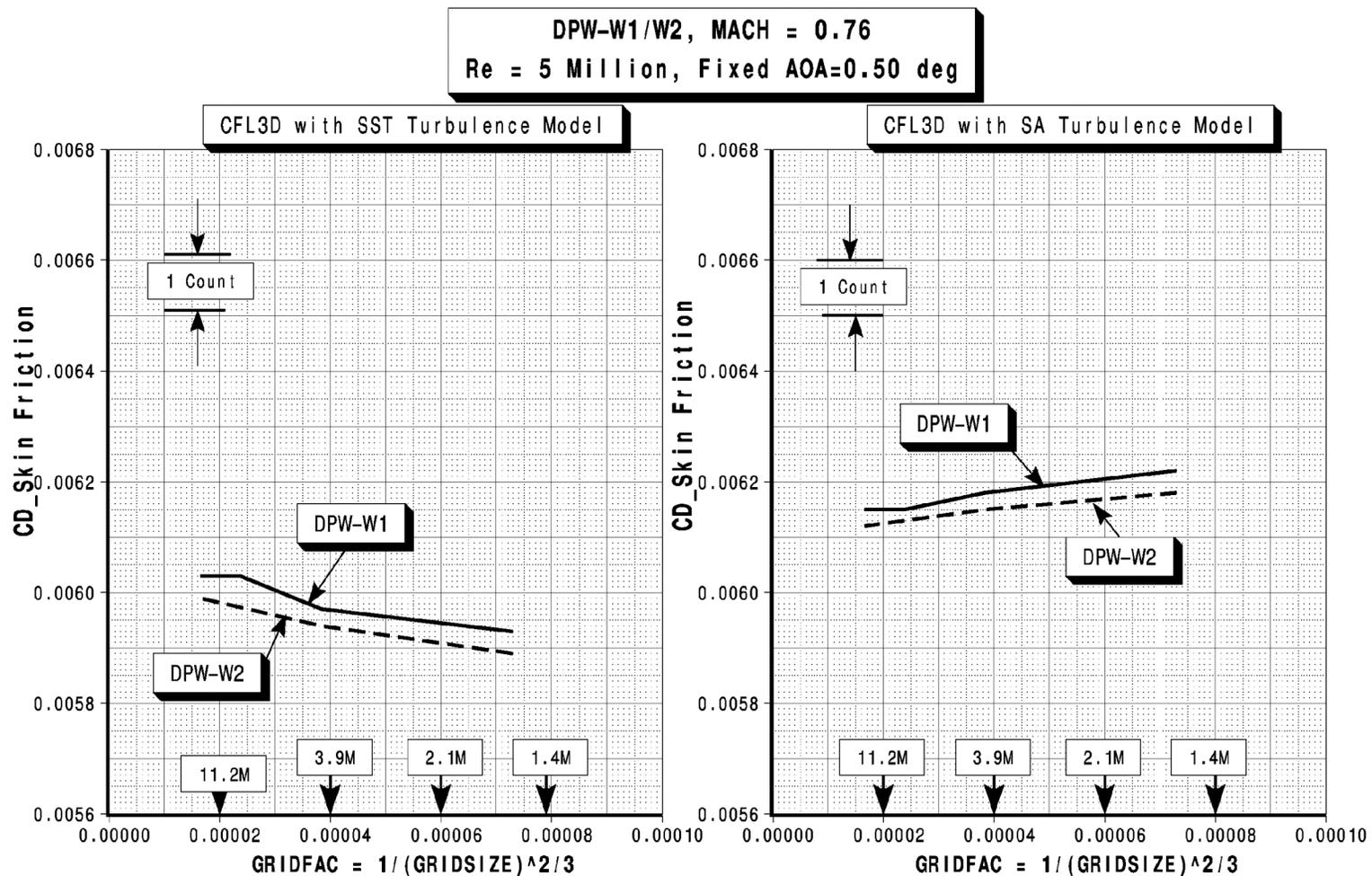
DPW-W1 / W2 – Drag Polar Increments



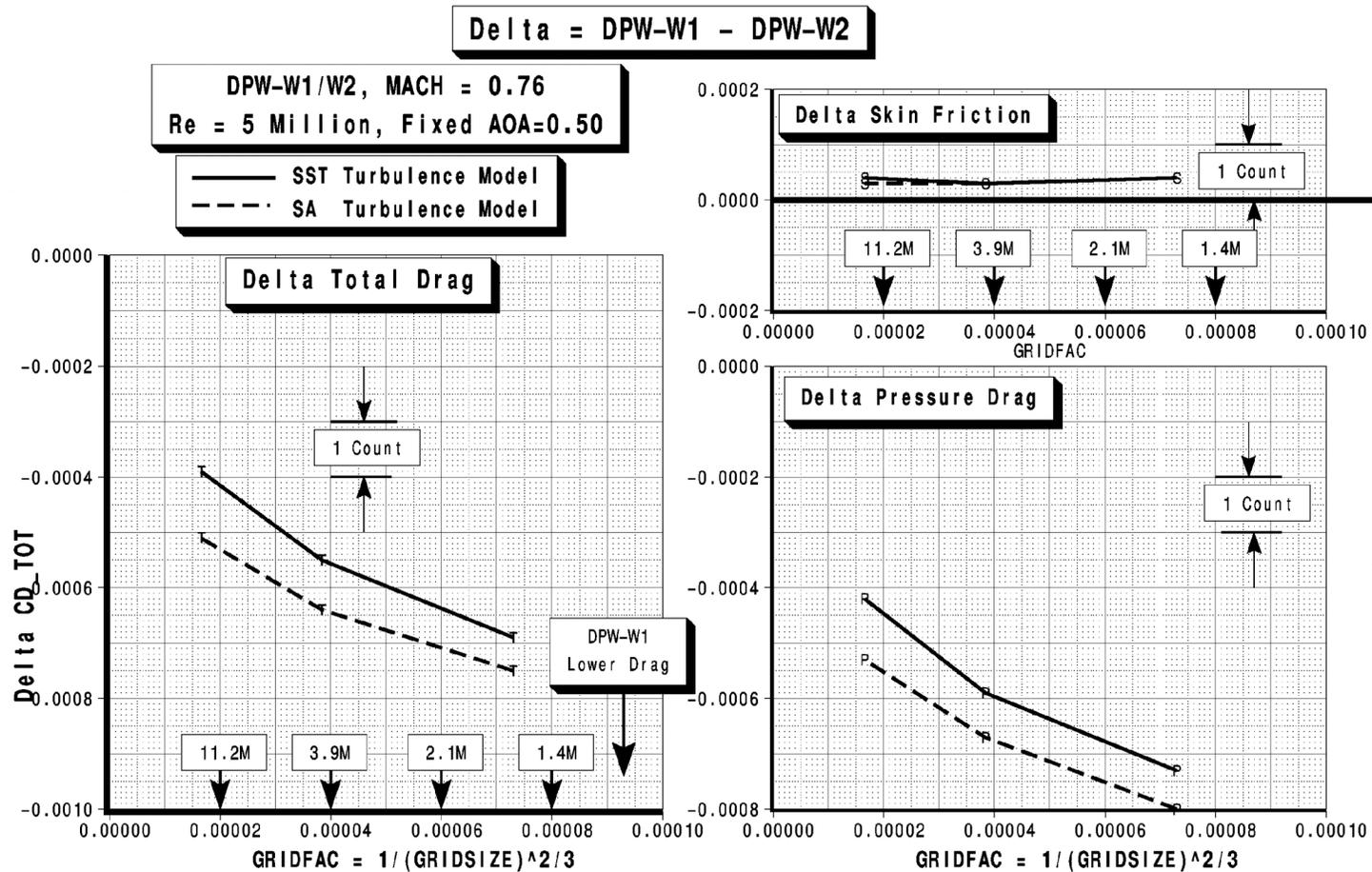
DPW-W1 / W2 – Total Drag Grid Convergence – CFL3D



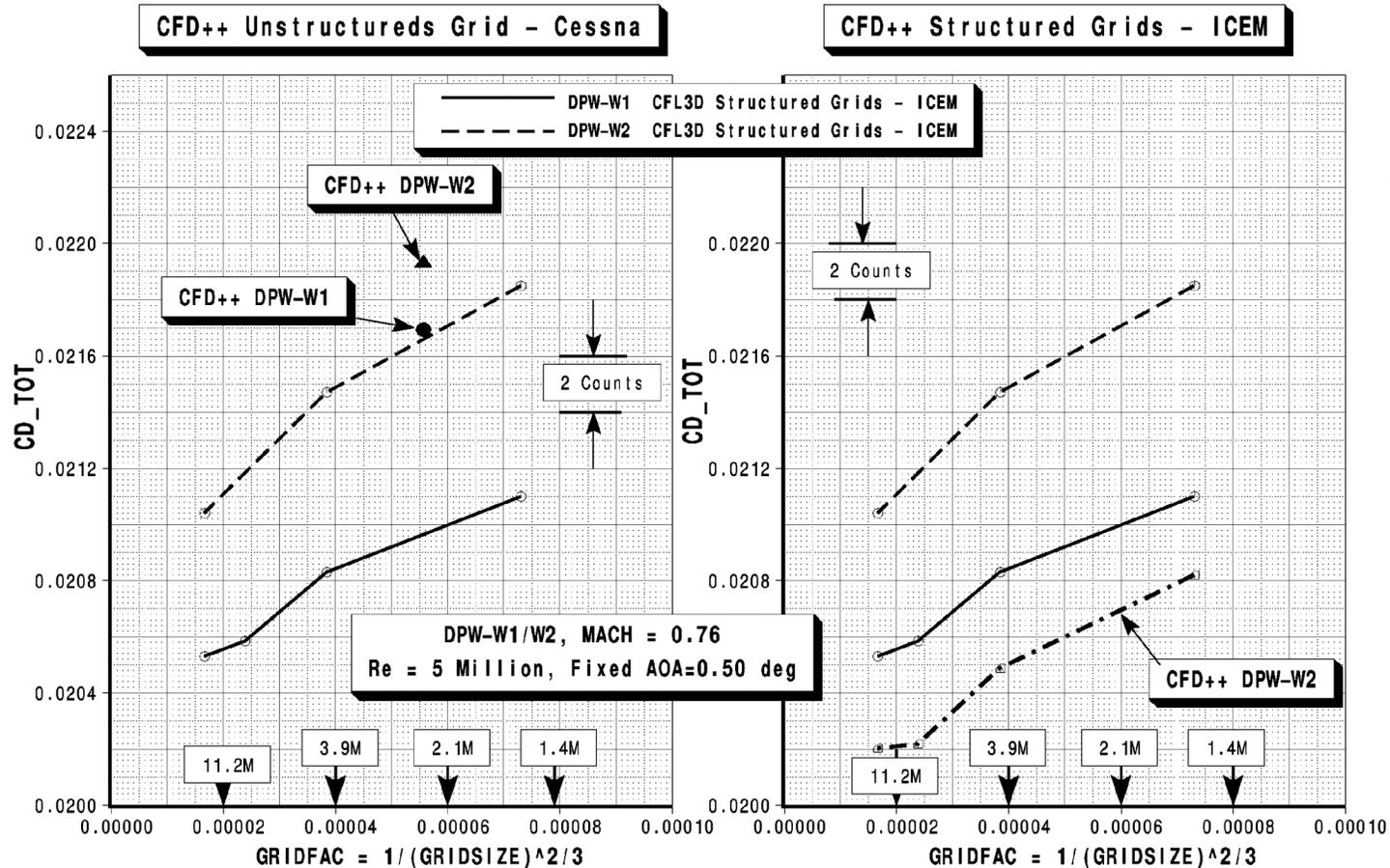
DPW-W1 / W2 – Skin Friction Grid Convergence – CFL3D



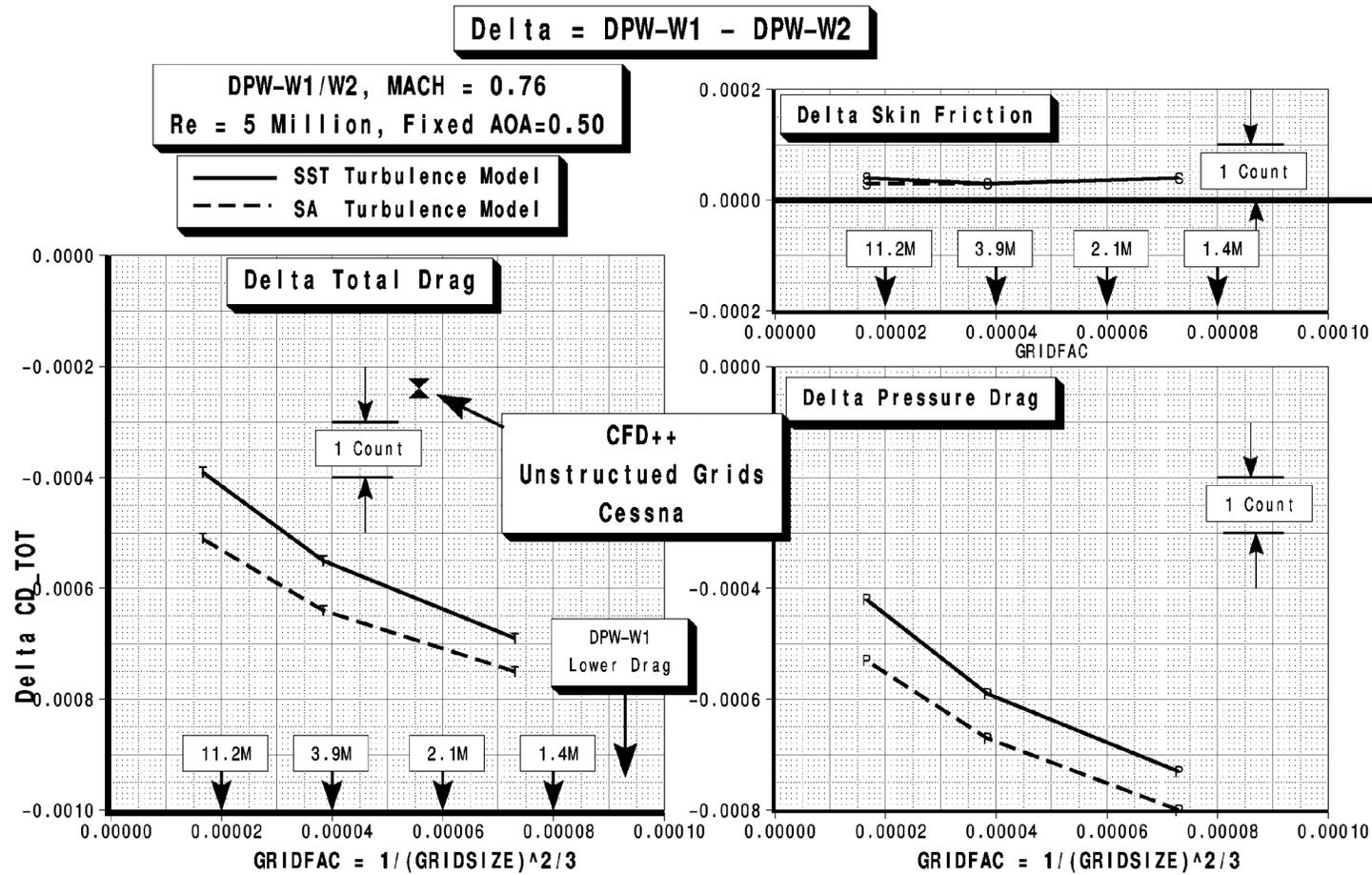
DPW-W1 / W2 – Drag Increment Grid Convergence – CFL3D



DPW-W1 / W2 – Total Drag Grid Convergence – CFD++



DPW-W1 / W2 – Drag Increment Grid Convergence – CFD++



Concluding Remarks

CFL3D – Structured ICEM Grids

- Convergence characteristics not as good as seen for F6 cases
- More variation in grids

CFD++ - Structured ICEM Grids

- Good convergence characteristics

CFD++ - Unstructured Cessna Grids

- No convergence problems on Cessna medium grid for W1.
- Divergence observed on all cases for W2 except for a =2 deg. Solution for all other angles obtained using this as a restart solution.

We should rename to the Grid Convergence Workshop



Applied Aerodynamics
Technical Committee

3rd CFD Drag Prediction Workshop

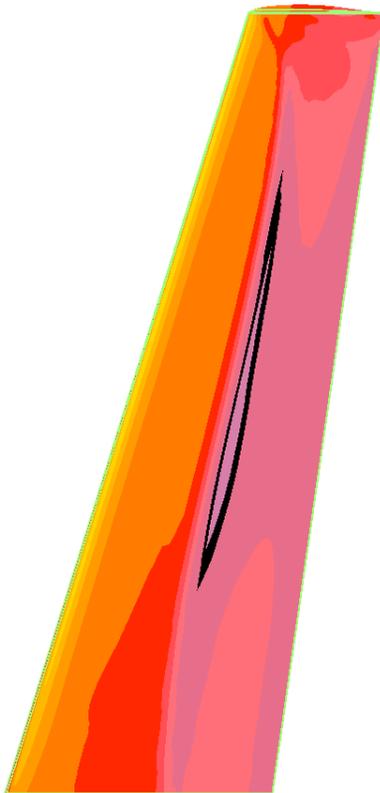
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Backup

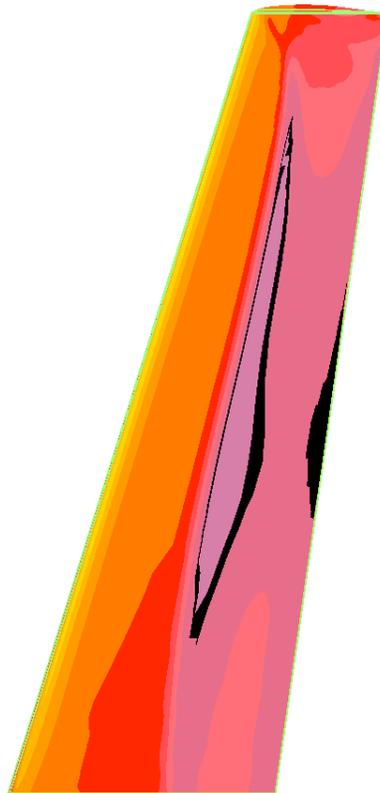
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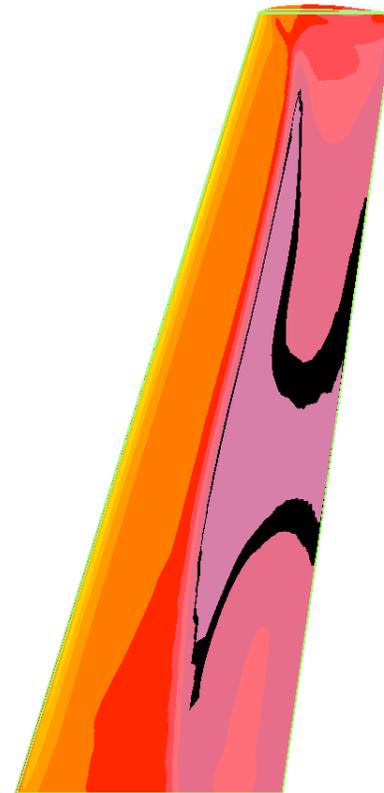
w2 - $\alpha=2.0$



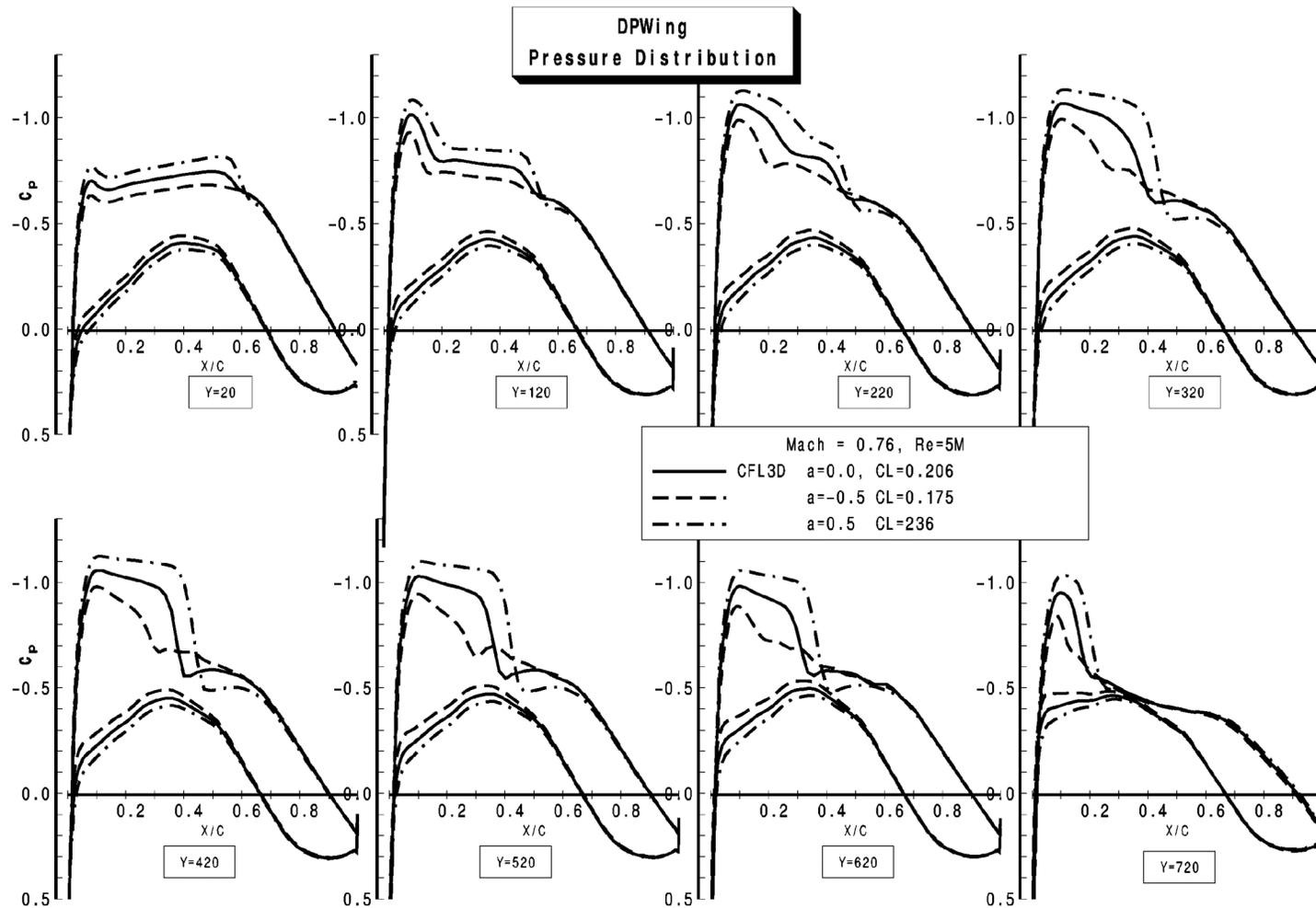
w2, $\alpha=2.5$



w2, $\alpha=3.0$

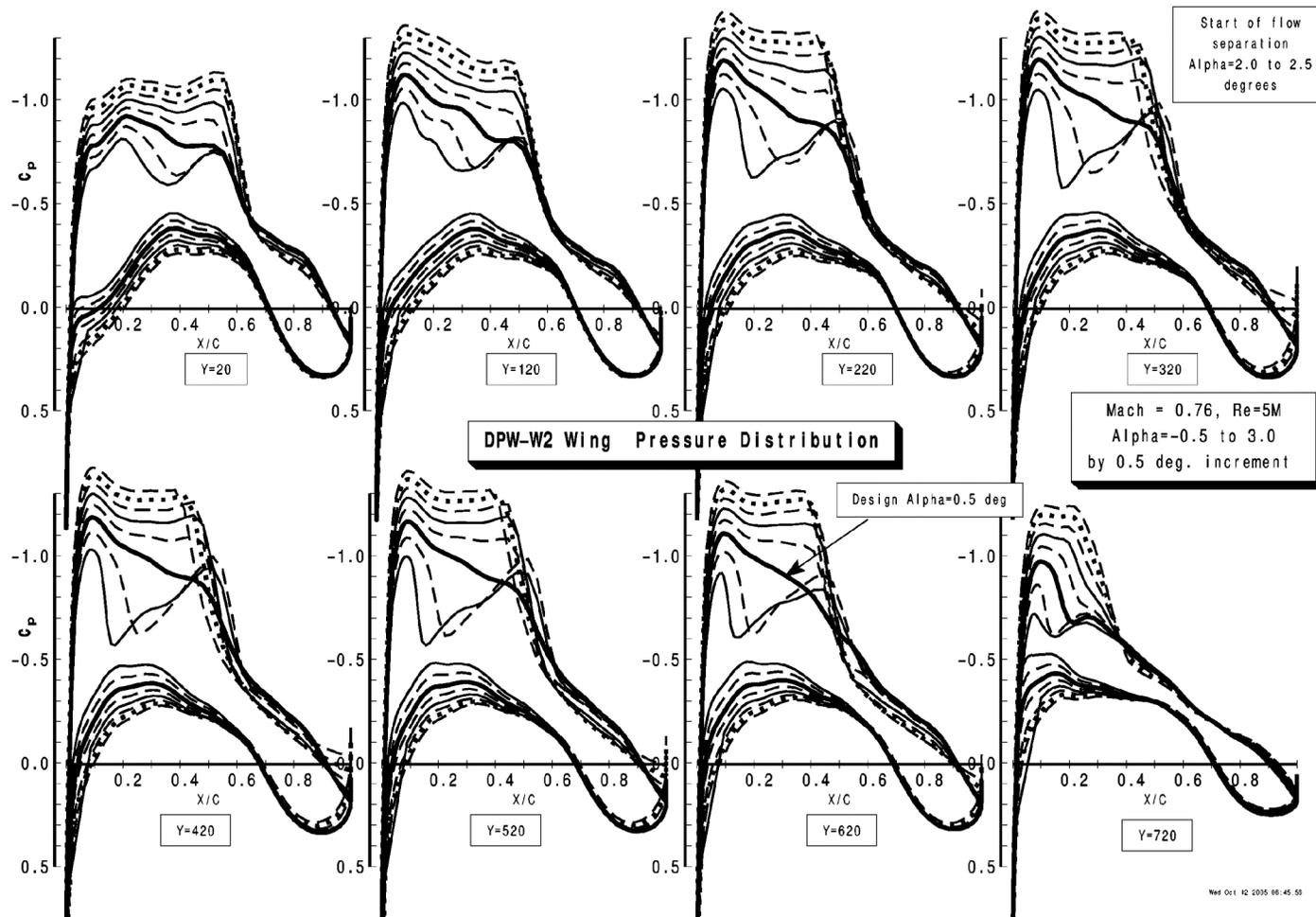


DPW-W1 Pressure Distributions



Tue May 5 2005 10:04:05

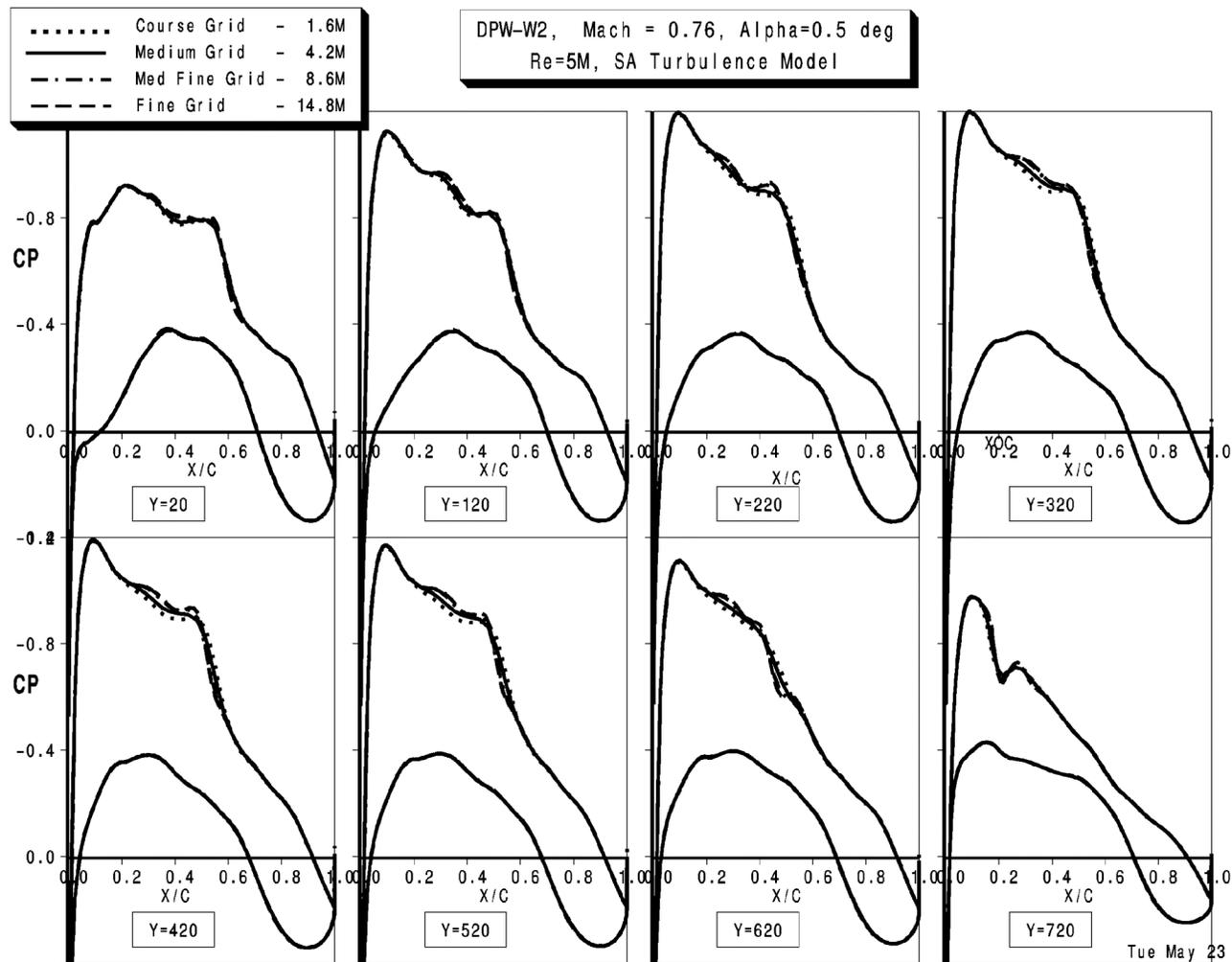
DPW-W2 Pressure Distributions





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Tue May 23